

# Resolution of Non-Linear Problems In Realistic-Lung-Inflating Simulation with Finite Element Method

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## ABSTRACT

Hadrontherapy treatment needs accurate tumour targeting, which is difficult for lung cancer due to breathing motions. We propose to quantify lung deformations and displacements by a simulation technique based on the geometrical and mechanical properties of organs. Thereby, we model lung behaviour by a 3D dynamic deformable model derived from continuous mechanics, computed with finite elements method (FEM).

## 1 Introduction

### 1.1 State of the Art

Various approaches have been explored to globally model lungs in 3D :

- Zordan *et al.* [2] developed a model of thorax breathing for animation purpose. Lung motions are driven by **spring-muscle elements**.
- Kaye *et al.* [3] studied the kinematics of lung with a **mass-spring system**, assuming a uniform external pressure.
- Amrani [4] has developed a model based on a **particle system** that undergoes deformations due to inflation and obstacles.
- Grimal *et al.* [5] used a **finite element model** to study thoracic impact injuries. Breathing motion was not included.

The state of the art shows that the existing models **do not use any personalised parameter** or **are not accurate enough** to properly track tumours.

### 1.2 Our context

- We model lung in the framework of continuum mechanics.
- Computation is solved with **FEM**.
- This method implies non-linearity such like:
  - Stiffness evolution according to **displacement**.
  - **Boundary condition**.

⇒ This method needs appropriated convergence parameters.

### 1.3 Questions

- How to model physiological and anatomical lung behaviour?
- How to integrate The patient's customised data?
- How to compute the non-linear behaviours?
- How to fix parameters to have accurate results?

## 2 Our approach in few points

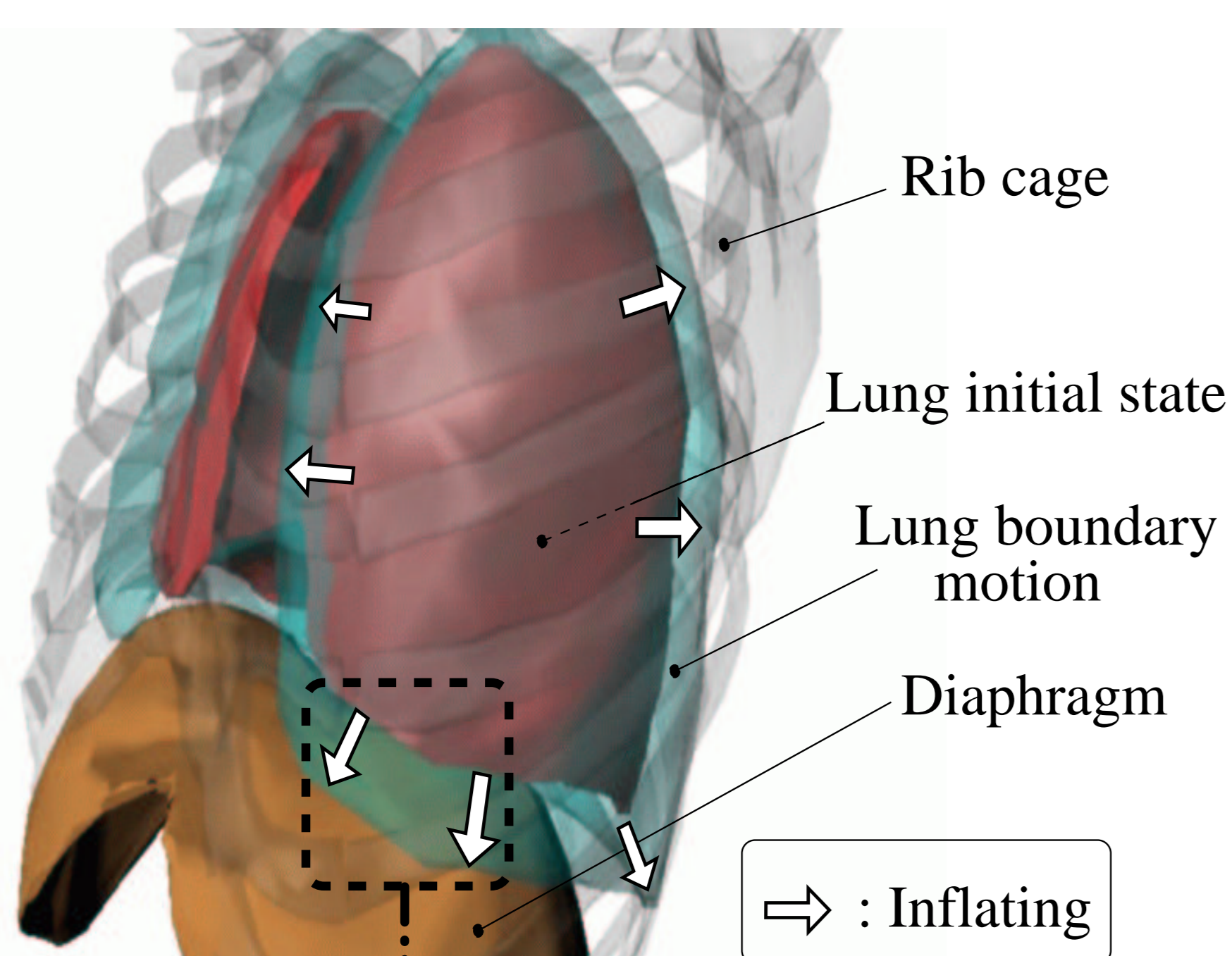
### 2.1 Lung Anatomy and Physiology

Lungs are wrapped in a fibrous membrane: the **pleura**, the very low pressure in the cavity making tissue remains in contact.  
⇒ Diaphragmatic and rib-cage-muscle actions cause pressure changes inside pleura.  
⇒ It steadily induces lungs inflating or deflating.

According to their location, Langen *et al.* [6] show that lung tumours can undergo large displacements.

### 2.2 Lung Inflation Modelling

- We simulate as precisely as possible the pleura behaviour:
  - The whole system is **fixed to the trachea**.
  - A **uniform negative pressure** is applied around the lung at its initial surface.
  - We check when the **external surface matches the external surface of the lung at its boundary motion**.
  - **Sliding without friction** (as pleura does) is allowed.
- ⇒ Surface to surface matching between two states.



## 3 Non-linearity problems

### 3.1 Finite Element Resolution

The FEM :

- is a numerical method;
- transforms the solution into a **matrix representation**;
- is based on space discretisation into small elements (**meshing**);
- consists here in **searching displacements  $U$  to reduce as possible the residue  $R$  defined by (1)**.

$$R(U) = F - K(U).U \quad (1)$$

where  $K$  is the **stiffness matrix** and  $F$  is the **load vector**. For large strains there is strong modifications in shape and  $F.K(U)$  is no linear.

This is the first source of non-linearity

The full system is then solved with the **Newton-Raphson algorithm**, an iterative method giving:

$$R(U) = R(U_{n-1}) + (U_n - U_{n-1}) \frac{\partial R}{\partial U}(U_{n-1}) \quad (2)$$

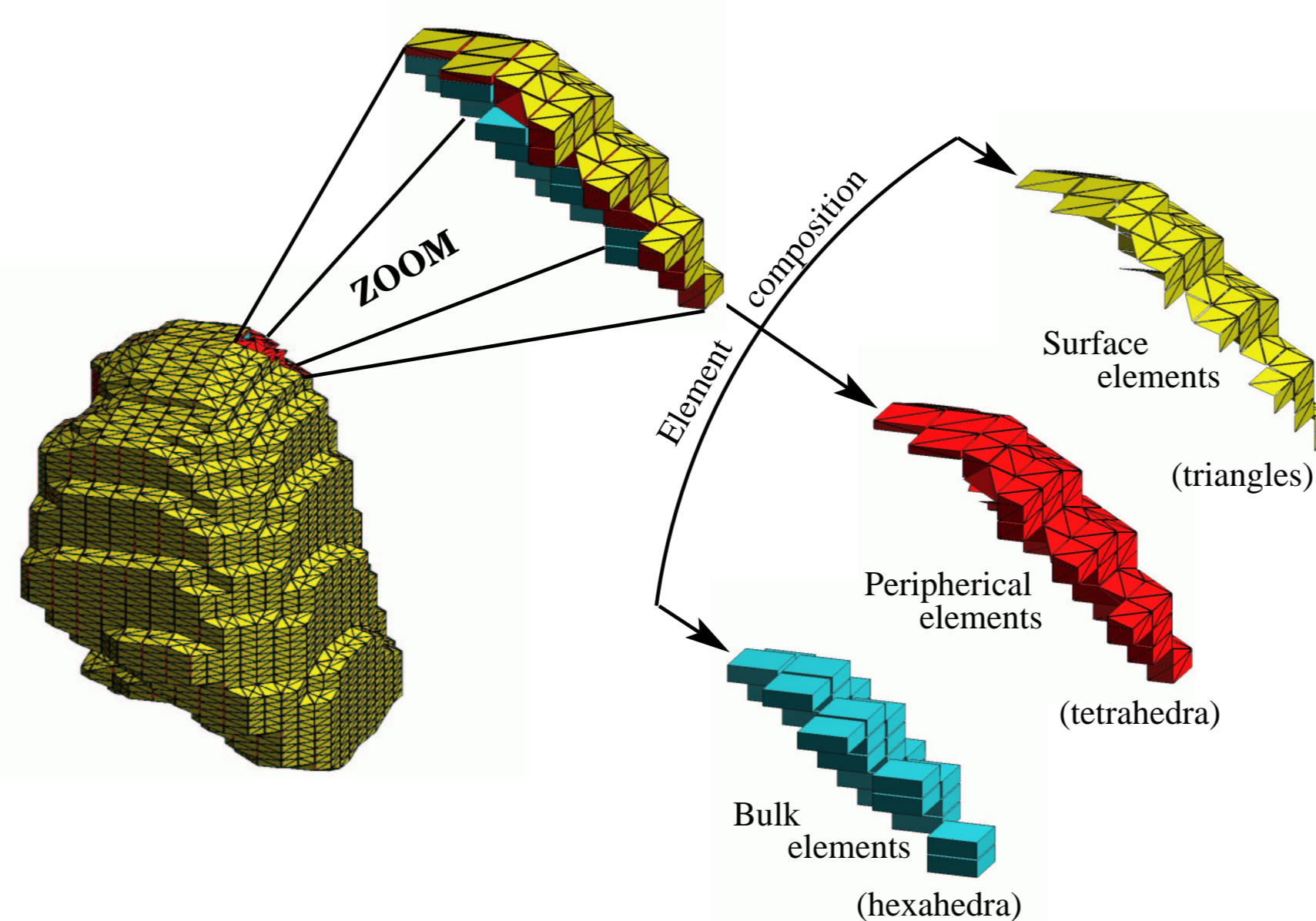
( $U_n$ ) is defined by :

$$\begin{cases} K(U_{n-1}).\Delta U_n - R(U_{n-1}) = 0 \\ \Delta U_n = U_n - U_{n-1} \end{cases} \quad (3)$$

### 3.2 Meshing

Convergence rate directly depends on mesh accuracy.

⇒ Mesh is divided into three entities: **triangles, tetrahedra and hexahedra**:



### 3.3 Large-Strain Problem

Lung volume increases by a factor of two during a respiration cycle.  
⇒ Large strains have then to be considered.  
⇒ we employ the algorithm presented by J.C. Simo *et al.* [7].

This method uses the **Cauchy-Green strain tensor**  $\epsilon_{cg}$  computed with the transformation gradient  $G$  of the geometrical deformation:

$$\epsilon_{cg} = 1/2(Id - (G.G^T)^{-1}) \quad (4)$$

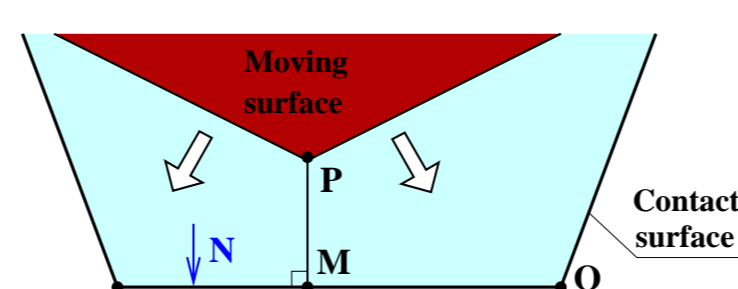
The stress tensor used is the **Kirchoff tensor**  $\tau(X_f)$ , computed like a "scaling" of the tensor  $\sigma(X_i)$ :

$$\tau(X_f) = det(G).\sigma(X_i) \quad (5)$$

where  $\sigma(X_i)$  is the state of stress at the position  $X_i$ .  
 $K(U_{n-1})$  is evaluated with (4) and (5), then the residue  $R(u_{n-1})$  is estimated according to (1), which gives  $U_n$  from (3).

### 3.4 Contact Problem

To handle contact conditions:



- Couples of points  $P$  and  $M$  potentially in contact are searched.
- Distance  $PM$  must be positive to satisfy the conditions of non penetration, *i.e.*:
 
$$PM_{n-1}.N + (U_M - U_P).N \leq 0 \quad (6)$$

• The equilibrium equation must be completed by a force to add compression and avoid a separation.

This is the second source of non-linearity

- An equation must be added to express that this force only takes part when contact is reached and only corresponds to a compression force.

If the imposed negative pressure (with respect to atmosphere) is not sufficiently important, residue  $R$  of equation (1) will be reached before contact condition.

⇒ The pressure value must then be large enough.

⇒ Convergence is ensured by sub-iterations taking into account geometry reactualisations.

## 4 Numerical experiments

### 4.1 Experiments Parameters

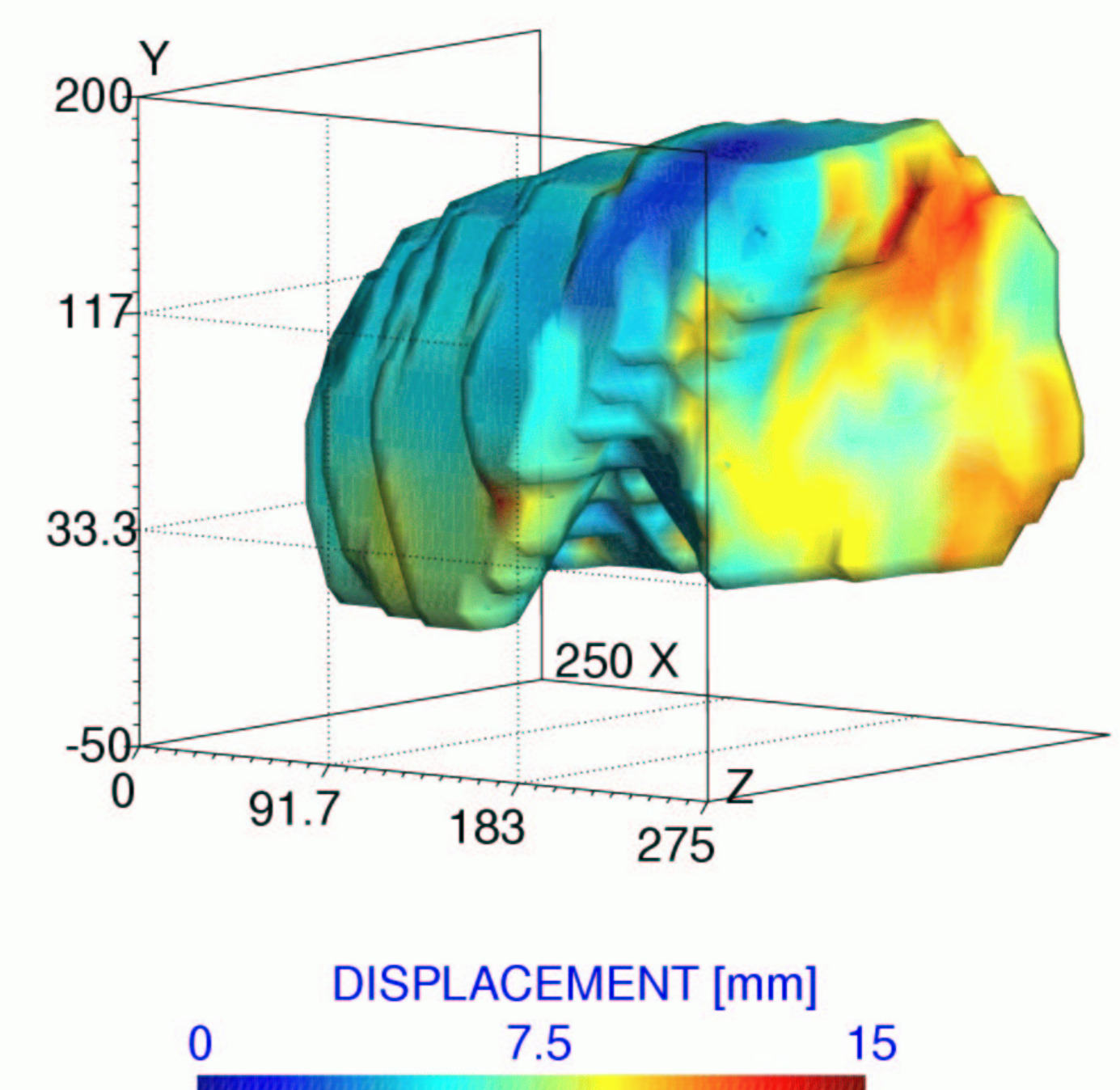
- Geometry is characterised by **10 778 points** and **30 097 elements**.
- Lung bounding box dimensions are **240 mm × 180 mm × 245 mm**.
- Measured patient compliance (lung elasticity) is **3.5 l/kPa**.
- Minimum residue of equation 3 is set to  **$R = 10^{-6}$** .
- Computation of lung motion have been calculated with several numbers of geometry reactualisation for the contact conditions.

Reactualisation number	Computing time	Average displacement <sup>α</sup> [mm]
1	14h43m	5.66
3	15h15m	5.23
5	15h28m	5.1
7	15h48m	5.1

We observed a correct convergence can be obtained after **5 iterations**.

<sup>α</sup>: Average displacement is computed with the **displacement-vector norm** of each mesh vertex.

### 4.2 Surface Deformation Results



- The displacement field is **totally smooth**.
- There is **no convergence problem** associated to mesh aberration.
- The **final surface matches** well with geometry extracted from CT scans (trial definition).

## 5 Conclusions

- A realistic way to model lung inflating is set.
- Convergence parameter related to non-linearity were managed.
- Results are convertible into 4D CT scan useful for physician and treatment planning soft.
- Do we need to relax any assumption (heterogeneity, anisotropy, ..) for hadrontherapy accuracy?

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